

Ans 1

DEPARTMENT OF CHEMICAL ENGINEERING
MODEL ANSWER
PROCESS EQUIPMENT DESIGN-I
AS-4169

Answers of multiple choice questions

(i) (a)

(ii) (c)

(iii) (b)

(iv) (d)

(v) (a)

(vi) (c)

(vii) (b)

(viii) (a)

(ix) (d)

(x) (a)

Solⁿ 2 :-

Given that :- A vessel

outside diameter (D_o) = 1.6 m

operating pressure (P) = 5 kg/cm²

permissible stress (f) = 1020 kg/cm²

welded joint efficiency = 85%

(9) Calculation of required thickness of vessel :-

$$t = \frac{PD_o}{2fj + P}$$

allowable pressure is taken 5% excess of operating pressure (p) = 5×1.05

$$p = 5.25 \text{ kg/cm}^2$$

Now, calculation of shell thickness (t) :-

$$t = \frac{5.25 \times 1.6}{2 \times 1020 \times 0.85 + 5.25}$$

$$t = 4.83 \times 10^{-3} \text{ m}$$

$$t = 4.83 \text{ mm}$$

2 mm corrosion allowance is assumed :-

$$t = 4.83 + 2 = 6.83 \text{ mm}$$

$t = 7 \text{ mm}$ with corrosion allowance.

For safe design, ~~Beckett~~ purpose, rechecking of answer will be based on ~~(t/D_i)~~ (t/D_i) ratio & maximum stress (F_r) (Resultant stress).

$$\frac{t}{D_i} = \frac{4.83}{1.6 \times 10^3} = 0.00303$$

$$F_r = 0.00303 \times 1020 = 3.0906 \text{ kg/cm}^2$$

from eqn (I) it is cleared that the $\left(\frac{t}{D_i}\right)$ ratio is much less than 0.25 (standard condition). Hence the thickness calculated is correct.

② Rechecking of answer based on stress:-

$$f_c = \frac{P(D_i + t)}{2t}$$

$$= \frac{5.25(1590.34 + 4.83)}{2 \times 4.83}$$

$$D_i = D_o - 2t$$

$$= 1.6 \times 10^3 - 2 \times 4.83$$

$$= 1590.34 \text{ mm}$$

longitudinal stress:-

$$(f_c) = \frac{P D_i}{4t}$$

$$\Rightarrow f_c = \frac{5.25 \times 1590.34}{4 \times 4.83}$$

$$\Rightarrow f_c = 432.15 \text{ kg/cm}^2$$

stress due to wt. of shell

$$f_w = \frac{w}{\pi t(D_i + t)}$$

$$\Rightarrow f_w = \frac{20000 \times 0.83}{\pi \times 4.83(4.83 + 1590.34)}$$

$$\Rightarrow f_w = 0.08266 \text{ kg/mm}^2$$

$$\Rightarrow f_w = 0.8109 \times 0.08266 \text{ kg/cm}^2$$

$$\Rightarrow f_w = 0.08266 \times 10^6 \text{ kg/cm}^2$$

$$f_w = 8.266 \text{ kg/cm}^2$$

$$f_a = f_c + (f_w) \Rightarrow f_c - f_w = 432.15 - 8.266$$

$$\Rightarrow f_a = 423.884 \text{ kg/cm}^2$$

Stress due to offset piping (torque is induced)

$$f_s = \frac{2T}{\pi t d_i (D_i + t)}$$

$$= \frac{2 \times 300 \times 10^3}{9.81 \times 3.14 \times 4.83 (1590.34 + 4.83)} \times 1590.34$$

$$= 1.5896 \text{ kg/mm}^2$$

$$f_s = 0.158 \text{ kg/cm}^2$$

$$f_R = [f_c^2 - f_a \times f_c + f_a^2 + 3f_s^2]^{1/2}$$

$$f_R = [866.94^2 - 423.884 \times 866.94 + 423.884^2 + 3 \times 0.158^2]^{1/2}$$

$$f_R = 750.85 \text{ kg/cm}^2$$

Permissible stress > f_R

$$1020 \text{ kg/cm}^2 > 750.85 \text{ kg/cm}^2 \quad \text{--- (II)}$$

Hence, the eq (I) & (II) improved the design is safe.

(b) Calculation of torispherical head.

Given:- for first approximation

$$R_i = D_o = R_o, \quad r_j = r_o = 6\% D_o$$

Outside diameter of shell (D_o) = 1.6 m

$$h_o = R_o - \left[\left(R_o - \frac{D_o}{2} \right) \times \left(R_o + \frac{D_o}{2} - 2r_o \right) \right]^{1/2} \quad \text{--- (1)}$$

$$h_o = 1.6 - \left[\left(1.6 - \frac{1.6}{2} \right) \times \left(1.6 + \frac{1.6}{2} - 2 \times 0.06 \times 1.6 \right) \right]^{1/2}$$

$$h_o = 1.6 - 1.329$$

$$h_o = 0.271 \text{ m}$$

$$\frac{D_o^2}{4R_o} = \frac{1.6^2}{4 \times 1.6}$$

$$\left[\frac{D_o^2}{4R_o} \right] = 0.4 \text{ m} \quad \text{--- (2)}$$

$$\left(\frac{D_o r_o}{2} \right)^{1/2} = \left(\frac{1.6 \times 0.06 \times 1.6}{2} \right)^{1/2}$$

$$\left(\frac{D_o r_o}{2} \right)^{1/2} = 0.277 \text{ m} \quad \text{--- (3)}$$

Out of three quantities calculated from eqn (1), (2) & (3). The least value from these three

eqn is $h_o = 0.271 \text{ m}$

Therefore,

Effective external height $h_E = h_o$

Now,
$$\frac{h_E}{D_o} = \frac{0.271}{1.6} = 0.16$$

(5)

Thickness of the head is determined from eqⁿ.

$$t = \frac{P D_0 C}{2 f J}$$

$$t = \frac{5.25 \times D_0 \times C}{2 \times 1020 \times 0.85}$$

$$\frac{t}{D_0} \times \frac{1}{C} = 0.00302 \quad \text{--- (4)}$$

$$\frac{h E}{D_0} = 0.16$$

$\frac{h E}{D_0}$	$\frac{t}{D_0}$	e	$\frac{1}{C}$	$\frac{t}{D_0} \times \frac{1}{C}$
0.16	0.002	3.8	0.263	0.002×0.263 $= 0.00052$
0.16	0.005	2.45	0.408	$= 0.00204$
0.16	0.01	1.95	0.512	$= 0.00512$
0.16	0.02	1.85	0.544	$= 0.01088$
0.16	$>> 0.04$	1.65	0.606	$= 0.02424$

Nearest value of eqⁿ (4)

hence,

$$\frac{t}{D_0} = 0.005$$

$$t = 0.005 \times 1.6$$

$$t = 8 \times 10^{-3} \text{ m}$$

$$t = 8 \text{ mm}$$

\Rightarrow

\Rightarrow

if 2mm Corrosion allowance is considered

$$t = 8 + 2$$

$$t = 10 \text{ mm}$$

Compare to outside diameter ($D_o = 1.6$), 10 mm is very small. Hence the first approximation does not introduce any considerable error in the result.

For next approximation

$$\begin{aligned} R_o &= R_i + t \\ &= 1.6 + 0.001 \\ &= 1.61 \end{aligned}$$

Then,

$$\begin{aligned} h_o &= 1.61 - \sqrt{\left(1.61 - \frac{1.6}{2}\right) \left(1.61 + \frac{1.6}{2} - 2 \times 0.06 \times 1.6\right)} \\ &= 1.61 - 1.34 \end{aligned}$$

$$h_o = 0.269 \text{ m}$$

$$\frac{D_o^2}{4R_o} = \frac{1.6^2}{4 \times 1.61} = 0.397 \text{ m}$$

$$\left(\frac{D_o r_o}{2}\right)^{1/2} = \left(\frac{1.6 \times 0.06 \times 1.6}{2}\right)^{1/2}$$

$$\left(\frac{D_o r_o}{2}\right)^{1/2} = 0.277$$

least value is $h_o = 0.269 \text{ m} \approx h_{cr}$

$$\frac{h_{cr}}{D_o} = \frac{0.269}{1.6} = 0.168$$

$$\frac{h_{cr}}{D_o} \approx 0.16$$

hence the second approximation also give the same

$$\frac{t}{D_o} = 0.005$$

$$t = 8 \text{ mm}$$

Thickness for torispherical head.

(7)

Soln. ③ :- Given that,

$$\text{Internal pressure } (P_i) = 140 \text{ MN/m}^2$$

$$\text{Internal diameter } (d_i) = 250 \text{ mm}$$

$$\text{Yield point of steel } (\sigma_y) = 480 \text{ MN/m}^2$$

$$\text{Factor of safety } (f_{os}) = 1.4$$

Wall thickness calculation by various failure theories.

① Maximum principal stress theory :-

$$\frac{\sigma_y}{(f_{os})} = P_i \left(\frac{k^2+1}{k^2-1} \right)$$

$$\frac{480}{1.4} = 140 \left(\frac{k^2+1}{k^2-1} \right)$$

$$\left(\frac{k^2+1}{k^2-1} \right) = \frac{480}{140 \times 1.4}$$

$$\left(\frac{k^2+1}{k^2-1} \right) = 2.19$$

$$(k^2+1) = (k^2-1) \times 2.19$$

$$k^2+1 = 2.19k^2 - 2.19$$

$$2.19k^2 - k^2 - 2.19 = 1 = 0$$

$$1.19k^2 - 3.19 = 0$$

$$k^2 = \frac{3.19}{1.19}$$

$$k^2 = 2.68$$

$$k = 1.637$$

\therefore f_{os} = factor of safety

We know that,

$$K = \frac{D_o}{D_i}$$

$$D_o = K D_i$$

$$= 1.687 \times 250$$

$$D_o = 409.32 \text{ mm}$$

hence, wall thickness will be,

$$t = \frac{D_o - D_i}{2}$$

$$t = \frac{409.32 - 250}{2}$$

$$t = 79.66 \text{ mm}$$

①

② Maximum strain theory:-

$$\frac{\sigma_y}{f_{os}} = P_i \left[\frac{(1-\mu) + (1+\mu)k^2}{(k^2-1)} \right]$$

Assume :- Poisson ratio for steel material
(μ) = 0.3

$$\frac{430}{1.4} = 140 \left[\frac{(1-0.3) + (1+0.3)k^2}{(k^2-1)} \right]$$

$$\Rightarrow \frac{430}{1.4 \times 140} = \left[\frac{0.7 + 1.3k^2}{(k^2-1)} \right]$$

$$\Rightarrow 2.19 = \left[\frac{0.7 + 1.3k^2}{(k^2-1)} \right]$$

③

$$\Rightarrow 2.19(k^2-1) = 0.7 + 1.3k^2$$

$$\Rightarrow 2.19k^2 - 2.19 = 0.7 + 1.3k^2$$

$$\Rightarrow 2.19k^2 - 1.3k^2 - 2.19 - 0.7 = 0$$

$$\Rightarrow 0.89k^2 - 2.89 = 0$$

$$\Rightarrow k^2 = \frac{2.89}{0.89}$$

$$\Rightarrow k = 1.802$$

Now,

$$k = \frac{D_o}{D_i} \Rightarrow k D_i = D_o$$

$$\Rightarrow 1.802 \times 250 = D_o$$

$$\Rightarrow 450.5 = D_o$$

$$D_o = 450.5 \text{ mm}$$

wall thickness based on 'maximum strain theory'

$$t = \frac{D_o - D_i}{2}$$

$$t = \frac{450.5 - 250}{2}$$

$$t = 100.25 \text{ mm}$$

— (2)

8) Maximum strain energy theory:-

$$\frac{\sigma_f}{(Fos)} = \frac{P_i \sqrt{6+10k^4}}{2(k^2-1)}$$

$$\frac{430}{1.4 \times 140} = \frac{\sqrt{6+10k^4}}{2(k^2-1)}$$

$$\Rightarrow 2.19 = \frac{\sqrt{6+10k^4}}{2(k^2-1)}$$

$$\Rightarrow 2(k^2-1) \times 2.19 = \sqrt{6+10k^4}$$

$$\Rightarrow 4.38k^2 - 4.38 = \sqrt{6+10k^4}$$

$$\Rightarrow (4.38k^2 - 4.38)^2 = 6 + 10k^4$$

$$\Rightarrow (4.38k^2)^2 - 2 \times 4.38k^2 \times 4.38 + 4.38^2 = 6 + 10k^4$$

$$\Rightarrow 19.18k^4 - 38.37k^2 + 19.18 = 6 + 10k^4$$

$$\Rightarrow 19.18k^4 - 10k^4 - 38.37k^2 + 19.18 - 6 = 0$$

$$\Rightarrow 9.18k^4 - 38.37k^2 + 13.18 = 0$$

$$k^2 = 3.802$$

$$\Rightarrow k = 1.949$$

Now,

$$k = \frac{D_o}{D_i} \Rightarrow k D_i = D_o$$

$$\Rightarrow 1.949 \times 250 = D_o$$

$$\Rightarrow 487.47 = D_o$$

$$D_o = 487.47 \text{ mm}$$

Wall thickness based on "Maximum strain energy theory"

$$t = \frac{D_o - D_i}{2}$$

\Rightarrow

$$t = \frac{487.47 - 250}{2}$$

\Rightarrow

$$t = 118.735 \text{ mm}$$

4) Maximum Shear Theory :-

$$\frac{\sigma_y}{(f_{os})} = \left(\frac{2k^2}{k^2-1} \right) P_i$$

\Rightarrow

$$\frac{430}{1.4 \times 140} = \left(\frac{2k^2}{k^2-1} \right)$$

\Rightarrow

$$(2.19)(k^2-1) = 2k^2$$

\Rightarrow

$$2.19k^2 = 2.19 - 2k^2 = 0$$

\Rightarrow

$$0.19k^2 - 2.19 = 0$$

\Rightarrow

$$k^2 = \frac{2.19}{0.19}$$

\Rightarrow

$$k = 3.395$$

Now,

$$k = \frac{D_o}{D_i} \Rightarrow k D_i = D_o$$

$$\Rightarrow 3.395 \times 250 = D_o$$

$$\Rightarrow 848.76 = D_o$$

$$D_o = 848.76 \text{ mm}$$

Thickness based on "Maximum shear theory"

$$t = \frac{D_o - D_i}{2}$$

$$t = \frac{248.76 - 250}{2}$$

$$t = 299.38 \text{ mm}$$

Solⁿ: (4)

Given data:- A fractionating tower

outside diameter of tower = 3000 mm
= 3 m

tangent to tangent length of tower = 5000 mm
= 5 m

Trey spacing = 1 m

$$R_i = D_o, \quad r_i = 0.1 D_o$$

$$f = 8.6 \times 10^6 \text{ kgf/m}^2$$

$$E = 19.9 \text{ kgf/mm}^2$$

$$E = 19.9 \times 10^6 \text{ kgf/m}^2$$

$$\text{Density of steel} = 7850 \text{ kg/m}^3$$

$$w/m = 10.4 \text{ kg/m}$$

① Determination of shell thickness without stiffness

Assume: At full vacuum at 400°C, pressure = 0.1 MN/m²

② ~~Assume~~ torispherical head used.

at initial approximation $D_o = D_i$ & $R_i = D_o$

$$h_i = R_i - \sqrt{\left(R_i - \frac{D_i}{2}\right) \times \left(R_i + \frac{D_i}{2} - 2r_i\right)}$$

$$\Rightarrow h_i = 3 - \sqrt{\left(3 - \frac{3}{2}\right) \times \left(3 + \frac{3}{2} - 2 \times 0.1 \times 3\right)}$$

$$\Rightarrow h_i = 3 - \sqrt{5.85}$$

$$\Rightarrow h_i = 3 - 2.41$$

$$\Rightarrow h_i = 0.58 \text{ m}$$

(2)

(14)

Effective length of the tower without stiffener.

$l_e =$ tangent to tangent length $+ \frac{2}{3}(h_i)$

$$l_e = 5 + \frac{2}{3}(0.58)$$

$$l_e = 5.38 \text{ m}$$

Therefore,

$$\frac{D_o}{l_e} = \frac{3}{5.38} = 0.55 \approx 0.6$$

\Rightarrow Based on $\left(\frac{D_o}{l_e}\right)$ value, get the value of K & m from code book.

$$K = 0.516$$

$$m = 2.49$$

① Shell thickness calculation for elastic stability:-

$$P = KE \left(\frac{t}{D_o}\right)^m$$

$$0.1 \times 10^6 = 0.516 \times 19.9 \times 10^6 \times 9.81 \left(\frac{t}{D_o}\right)^{2.49}$$

$$\frac{0.1}{0.516 \times 19.9 \times 9.81} = \left(\frac{t}{3}\right)^{2.49}$$

$$9.927 \times 10^4 = \left(\frac{t}{3}\right)^{2.49}$$

$$0.0622 = \left(\frac{t}{3}\right)$$

$$t = 0.186 \text{ m}$$

$$t = 186 \text{ mm}$$

⇒ checking for plastic deformation

$$P = 2 \times f \left(\frac{t}{D_0} \right) \times \frac{1}{1 + 1.5U(1 - 0.2 \frac{D_0}{L_e})} \times \frac{100 (t/D_0)}{100 (t/D_0)}$$

Assume: out of roundness for new vessels.
 $U = 1.5$

$$P = 2 \times 8.6 \times 10^6 \times 9.81 \left(\frac{0.126}{3} \right) \times \frac{1}{1 + 1.5 \times 1.5 (1 - 0.2 \times \frac{5.38}{5.38})} \times \frac{1}{100 \left(\frac{0.126}{3} \right)}$$

$$\Rightarrow P = 10461384 \times \frac{1}{1 + \frac{1.98}{6.2}}$$

$$\Rightarrow P = 10461384 \times 0.758$$

$$\Rightarrow P = 7931299.4 \text{ N/m}^2$$

$$\Rightarrow P = 7.93 \text{ MN/m}^2$$

Since $7.93 \text{ MN/m}^2 > 0.1 \text{ MN/m}^2$

The calculated thickness is safe against plastic deformation.

②. If stiffeners are used, the effective length of the tower will be reduced to 1 m (tray spacing)

$$L = 1$$

$$D_0 = \frac{3}{3} = 3$$

$$= 4949472 \times 0.765$$

$$= 3786346.08$$

$$= 3.78 \text{ MN/m}^2$$

$$3.78 \text{ MN/m}^2 > 0.1 \text{ MN/m}^2$$

Therefore the calculated thickness is sufficient for plastic deformation.

(b) Design of stiffening ring involves first select a standard ^{structure} ring then to check for required moment of Inertia calculated.

$$I_{\text{req}} = \frac{D_o^2 L \left(t + \frac{A_s}{L} \right) f}{12 \times E}$$

$$= \frac{3^2 \times 1 \left(0.088 + \frac{A_s}{1} \right) \times 8.6 \times 10^6}{12 \times 19.9 \times 10^6}$$

$$I_{\text{req}} = 0.324 (0.088 + A_s) \text{ m}^4$$

$$= 0.324 (0.088 + A_s) \text{ m}^4$$

from chart (code book)

$$I_{\text{same}} \rightarrow \text{max. sectional area} = 62.93 \text{ cm}^2$$

$$= 62.93 \times 10^{-4} \text{ m}^2$$

$$I_{\text{req}} = 0.324 (0.088 + 0.006293)$$

$$= 0.030 \text{ m}^4$$

$$\text{But } I_{\text{actual}} = 0.0151 \text{ m}^4$$

hence

$$I_{req} > I_{actual}$$

Not satisfied the safe design condition
Therefore the correct value of I_{req} can be
obtained from Claret (Codebook - IS-2825-1969)
(Appendix f)

$$\frac{L}{D_0} \text{ at } 400^\circ\text{C}$$

$$\frac{L}{D_0} = \frac{4}{3} \text{ at } 400^\circ\text{C}$$

$$\frac{L}{D_0} = 0.33$$

$$\text{factor } A = 0.000026$$
$$f/E = 0.000026$$

from appendix F, P22
Fig. F.2 (IS-2825-1969)

Now,

Assume $A_s = 11.70 \text{ cm}^2$

$$I_{req} = \frac{3^2 \times 1 (0.088 + 0.000170) \times 0.000026}{12}$$

$$I_{req} = 1.738 \times 10^{-6} \text{ m}^4$$

$$I_{actual} = 186.7 \text{ cm}^4$$

$$= 186.7 \times 10^{-8}$$

$$= 1.86 \times 10^{-6} \text{ m}^4$$

Now,

$$I_{req} < I_{actual}$$

No. of stiffeners required (n) = tangent to tangent $k_g/4$
- tray spacing

$$\Rightarrow n = 5 - 1$$

$$\Rightarrow n = 4$$

$$\begin{aligned} \text{Total wt. of strgs} &= n \pi D_o \times w/m \\ &= 4 \times 3.14 \times 3 \times 10.4 \\ &= 391.272 \text{ kg} \end{aligned}$$

Saving the shell material for using stiffening strgs,

$$(W) = \pi D_o (t_{\text{min}} - t_1) \times L \times S_3$$

where, t_2 = shell thickness without stiffeners

t_1 = shell thickness with stiffeners

$$\text{so, } W = 3.14 \times 3 (0.186 - 0.088) \times 5 \times 7850$$

$$W = 36234 \text{ kg}$$

Therefore, use of stiffening strgs will be advantageous in this case.

Note:-

This problem may also solve by Alternative method using ^{Cheut-from} Appendices F

IS 2825-1969 and table 3.1 from code book (IS-2825-1969)

Given that,

$$\text{Density of process fluid } (\rho_f) = 1050 \text{ kg/m}^3$$
$$\text{plate size} = 0.9 \text{ m} \times 1.8 \text{ m}$$

$$\text{Density of steel } (\rho_s) = 7800 \text{ kg/m}^3$$
$$\text{tank closed by} = \text{flat covers.}$$

$$\text{max. diameter of tank } (D) = 2.6 \text{ m}$$
$$\text{the length of tank } (L) = 5 \text{ m}$$

$$\text{Thickness of the plate for shell fabrication} = 6 \text{ mm}$$

$$\text{Thickness of plate for closure} = 8 \text{ mm}$$

$$\text{Capacity of storage tank } (V_{\text{given}}) = 25 \text{ m}^3$$

Now,

$$\text{Cross sectional area of tank} = \frac{\pi (D)^2}{4}$$

$$(A_t) = \frac{3.142 (2.6)^2}{4}$$

$$(A_t) = 5.309 \text{ m}^2 \quad \text{--- (1)}$$

$$\text{Volume of storage tank} = (A_t) \times (\text{length of tank})$$

$$(V_t) = 5.309 \times 5$$

$$(V_t) = 26.545 \text{ m}^3 \quad \text{--- (2)}$$

\% Excess volume available in storage tank

$$= \left(\frac{V_t - V_{\text{given}}}{V_{\text{given}}} \right) \times 100$$

$$= \left(\frac{26.545 - 25}{25} \right) \times 100$$

$$\% \text{ Excess volume} = 6.18\%$$

Circumference of shell of a tank

$$\begin{aligned} &= \pi D \\ &= 3.142 \times 2.6 \\ &= 8.1692 \text{ m} \end{aligned}$$

Given 6mm thick shell plate \therefore

weight of process fluid = $25 \times S_f$

$$(W) = 25 \times 1050$$

$$(W) = 26250 \text{ kg}$$

Surface area of shell = πDL

$$= 3.142 \times 2.6 \times 5$$

$$\text{Surface area of shell} = 40.846 \text{ m}^2$$

Now, weight of one m^2 of shell can be calculated

as, $= (\text{Area of shell}) \times \text{shell thickness} \times \rho_{\text{shell}}$

$$= 1.0 \times 1.0 \times 0.006 \times 7800$$

$$= 46.8 \text{ kg} \Rightarrow \text{wt}/\text{m}^2 = 46.8 \text{ kg/m}^2$$

total weight of shell = ~~40.846~~

= Surface Area of shell \times wt of $\frac{1 \text{ m}^2 \text{ shell}}$

$$= 40.846 \times 46.8$$

$$\text{total wt. of shell} = 1911.6 \text{ kg}$$

∴ for the closing the shell, 8 mm thickness plate is used,

So,

$$\begin{aligned}\text{Surface area of both cover plates} &= \frac{\pi}{4} (D)^2 \times 2 \\ &= \frac{\pi}{4} (2.6)^2 \times 2 \\ &= 10.61 \text{ m}^2\end{aligned}$$

Then, weight of one m^2 , 8 mm thick plate will be

$$\begin{aligned}&= 1.0 \times 1.0 \times 0.008 \times 7800 \\ &= 62.4 \text{ kg}\end{aligned}$$

Total weight of cover plates

$$\begin{aligned}&= \text{Surface area of both cover plates} \times (\text{weight of } 1 \text{ m}^2 \text{ 8 mm thick plate}) \\ &= 10.61 \times 62.4 \\ &= 662.064 \text{ kg}\end{aligned}$$

Now,

total weight of empty tank along with cover plates = (total wt. of shell) + (total wt. of cover plates)

$$W' = 1911.6 + 662.064$$

$$W' = 2573.664 \text{ kg}$$

$$W = 22574 \text{ kg} \quad \text{--- (6)}$$

Effective span between the support $l_1 = \frac{L}{2} = \frac{5}{2} = 2.5 \text{ m}$

Maximum tensile bending stress in the shell plate

$$f = \frac{D(W+w)d_1^2 \times 100}{8L \times 0.098(D_1^4 d^4)} \quad \text{kg/cm}^2$$

∴ outside diameter of tank = $1D + 2 \times \text{thickness}$
 $= 260 + 2 \times 60$
 $= 380 \text{ cm}$

$$f = \frac{261.2(26250 + 2574) \times 250 \times 1}{8 \times 500(0.098)(261.2^4 \times 260^4)}$$

$$f = 14.13 \text{ kg/cm}^2 \quad \text{--- (7)}$$

maximum tensile stress in shell plate is negligible.

$$\begin{aligned} \text{maximum shear force} &= \frac{(W+w')L}{2L} \\ &= \frac{(26250 + 2574) \times 2.5}{2 \times 500} \\ &= 7206 \text{ kg} \end{aligned}$$

maximum direct stress = $\frac{\text{max. Shear force}}{\text{Cross sectional area of hollow tank}}$

$$\begin{aligned} &= \frac{7206 \times 4}{\pi(D^2 - d^2)} \\ &= \frac{7206 \times 4}{3.14(261.2^2 - 260^2)} \end{aligned}$$

$$\text{maximum direct stress} = 14.67 \text{ kg/cm}^2 \quad \text{--- (8)}$$

total fluid pressure on each cover plate, when the tank is full.

$$(\text{Wt of fluid on each cover plate}) = S_f \times A_p \times H/2$$

$$= \frac{1050 \times \pi \times 2.6^2 \times H}{4}$$

where $A_p =$ cross sectional area of plate

$$= \frac{1050 \times \pi \times 2.6 \times 1.3}{4} \quad H = 0.87$$

$$= 6964.9 \text{ kg}$$

$$\approx 7244 \text{ kg} \quad \text{--- (9)}$$

Wt of fluid on each cover plate

The pressure is assumed as uniformly distributed on the surface area of the cover plates,

Therefore,

$$\text{Avg. Pressure } P = \frac{\text{Wt of fluid on each cover plate}}{\text{Cross sectional area of cover plate}}$$

$$= \frac{7244 \times 4}{3.14 (260)^2}$$

$$\text{Avg Press. (P)} = 0.1365 \text{ kg/cm}^2$$

So, maximum stress on cover plate in bending

$$f = \frac{3}{16} \frac{P d^2}{t_{\text{plate}}^2} = \frac{3}{16} \times \left(\frac{0.1365 \times 260^2}{0.8^2} \right)$$

$$= 2703.3 \text{ kg/cm}^2$$

$$f = 2703.3 \text{ kg/cm}^2 \quad \text{--- (10)}$$

Since the stress on cover plate (eqⁿ (10)) is very high.

Therefore 10mm cover plate thickness is used

$$f = \frac{3}{16} \times \left(\frac{0.1365 \times 260^2}{(4.0)^2} \right)$$

$$f = 1730 \text{ kg/cm}^2 \quad \text{--- (11)}$$

~~Since~~ from eqⁿ (11) we found that the stress value is still high,

so provide, vertical angle stiffeners of 75x75x6mm and

The angle size of 75x75x6mm.

(from steel table.)